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This invention concerns a thrust nozzle for a gas turbine engine.

More specifically, the invention is concerned with a thrust nozzle, the outlet area of which can be varied, so as to cater for varying volumes of gas flow from a gas turbine engine which includes the nozzles.

The invention seeks to provide a variable area nozzle structure suitable for use as inter alia, a thrust vectoring nozzle i.e. one which is rotatable so as to emit propulsive gas flow in various directions.

According to the present invention, there is provided a variable area thrust nozzle comprising a tubular structure, <sup>curved in cross-section</sup> having an exhaust gas outlet, a flap mounted by one end, within the tubular structure about an axis which lies across said outlet, for pivoting movement about said axis between a position wherein said flap slopes towards a wall of said tubular structure and has its edges in sealing contact therewith and, a position wherein said flap lies parallel with said walls and moving means for achieving said movement.

The tubular structure may be circular in cross section.

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Preferably the tubular structure comprises an elbow having that end remote from the exhaust gas outlet, adapted for connection to gas turbine engine structure for rotation relative thereto.

The invention will now be described, by way of example and with reference to the accompanying drawings in which:

Figure 1 is a diagrammatic view of a gas turbine engine incorporating a pair of variable area thrust nozzles in accordance with an embodiment of the invention.

Figure 2 is an enlarged, cross-sectional part view of Figure 1.

In Figure 1 an aircraft power plant 10 comprises a core gas generator 12 driving a front fan 14. The fan duct 16 of front fan 14 terminates in a plenum chamber 18. The plenum chamber is divided into two portions 20, 22 each of which terminates in a respective propulsion nozzle 24, 26.

Propulsion nozzles 24, 26 are rotatable relative to the power plant 10, so as to enable thrust vectoring. In each nozzle, rotation is achieved by connecting the nozzles 24, 26 to the divided portions 20, 22 of plenum chamber 18, via bearings in respective housings 28, 30.

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A ram mechanism 56 lies along the exterior of nozzle 52 and is connected to flap 54 via a link 58.

5 Actuation of ram 56 pivots link 58 which in turn pivots flap 54 into the "reheat on" position.

10 As link 58 will pivot about one portion of an arc and flap 54 will pivot about another portion of arc, the connection between them will have to comprise a pin and slot connection, to cater for the different movements.